

QUALIFICATION ENVIRONMENTS

FOR

HS601 BIPROPELLANT TANK

ATK P/N 80350-1

Note: Not completed or finished.

Table 1: P/N 80350-1 HS601 BIPROPELLANT TANK

Assembly Specifications

Parameters	Requirements
Operating Pressure	psig
Proof Pressure	325 psig, Actual Proof: 325 psig
Burst Pressure	410 psid, Actual Burst: - psig, Rupture @ 555 psig
External Pressure	Not Tested
Internal Vacuum	Not Tested
Material of Construction	Pressure vessel constructed of 6AL-4V titanium.
Membrane Thickness	"
Tank Mount(s)	
Expulsion Efficiency	99.8%
Design Fill Fraction	-
Tank Capacity	22580 in ³
Internal Dimensions	42" Ø
Tank Weight	Maximum tank weight is 28.9 lbs, Actual tank weight is 26.98 lbs
Propellant Capacity	-
Shell Leakage	<1x10 ⁻⁶ std cc/sec He max, Actual: 9.2x10 ⁻⁹ scc/sec He @ 270 psig
Failure Mode	N/A
Natural Frequency	-
Temperature Environment	-
On Orbit Life	-

80350-1 was subjected to the following qualification tests:

<u>Test Sequence</u>	<u>Test Description</u>	<u>Pr</u> <u>1</u>
1.	Inspection	
2.	Tank Capacity	
3.	Proof Pressure Test	
4.	Tank Capacity	
5.	PMD Functional Test	
6.	External Leakage	
7.	Penetrant Inspection	
8.	Radiographic Inspection	
9.	Radiographic Inspection of PMD	
10.	Weight	
11.	Inspection and Powder Blast	
12.	Pressure Cycle	
13.	External Leakage	
14.	Acoustic Test	
15.	Sinusoidal Vibration	
16.	PMD Functional Test	
17.	Radiographic Inspection (Shell)	
18.	Radiographic Inspection of PMD	
19.	Random Vibration	
20.	External Leakage	
21.	PMD Functional Tests	
22.	Radiographic Inspection (Shell)	
23.	Radiographic Inspection of PMD	
24.	Collapse Pressure	
25.	External Leakage	
26.	Penetrant Inspection	
27.	Burst Pressure Test	

Note: The following tests are only listed in this report.

- 1) Pressure Log
- 2) Proof Pressure Test

Pressure Log

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TRW PRESSURE SYSTEMS, INC.

TRW PSI Procedure No. 50-000374
Page 33 Revision "N/C"

DATA SHEET "K" PRESSURE CYCLE LOG

Customer: Hughes Aircraft Company Date: MAY 16, 1990
Customer Part No. 4946407 TRW PSI Part No. 80350-1
Test Para. No. 3.6.1 TRW PSI Serial No. 0003
Test Equipment: SEE ATTACHED DATA SHEETS

Date	Pressure Event	Test Media	Pressure	Duration	Cycles	Inspector
5-16-90	PROOF	H ₂ O	325 psig	5min	1	
6-6-90	EXT. LEAK	HE	270 PSIG	30min	1	PSI 103
_____ _____ _____ _____ _____ _____ _____ _____ _____ _____ _____ _____ _____ _____ _____						

Tested By Martin Fucina Date 6-6-90
Witnessed By N/A  Date _____
WAC Witnessed By _____ Date _____

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TRW PRESSURE SYSTEMS, INC.

TRW PSI Procedure No. 50-000375
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DATA SHEET "W"
PRESSURE CYCLE LOG

Customer: Hughes Aircraft Company Date: 8-7-90

Customer Part No. 4946407 TRW PSI Part No. 80350-1

Test Para. No. 3.6.1 TRW PSI Serial No. 0003

Test Equipment: SEE RESPECTIVE DATA SHEETS

Date	Pressure Event	Test Media	Pressure	Duration	Cycles	Inspector
8/2/90	EXTERNAL LEAK (4.20)	He	270 PSIG	30.5 MIN	1	(PSI 0/8)
8/6/90	PRESSURE CYCLING	H ₂ O	325/260 PSIG	5 SEC MAX	2/15	(PSI 0/8)
8-7-90	EXT. LEAK (4.7)	He	270 PSIG	20 MIN	TWO	(PSI 0/8)
8-8-90	EXT. LEAK (4.7)	He	271 PSIG	30 MIN	ONE	(PSI 0/8)
9-13-90	EXT. LEAK (4.75)	He	270 PSIG	30 MIN	ONE	(PSI 0/8)
7/26/90	VIBRATION	FREON 113/EE/GN2	139 PSIG	41 HRS	3	(PSI 0/8)

Tested By [Signature] Date 9/13/90

Witnessed By [Signature] Date 9/13/90

HAC Witnessed By off-site Date 9/13/90
not required

40)

Proof Pressure Test

Tank is pressurized to 325 psig.

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TRW PRESSURE SYSTEMS, INC.

TRW PSI Procedure No. 50-000374
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DATA SHEET "B"
PROOF PRESSURE & INTERNAL VOLUME
(SHEET 1 OF 2)

Customer: Hughes Aircraft Company

Date: MAY 16, 1990

Customer Part No. 4946407

TRW PSI Part No. 80350-1

Test Para. No. 4.2.1, 4.2.2

TRW PSI Serial No. 0003

Test Equipment: E-0598 CAL. 4-18-90 DUE 10-18-90: ST-0625

CAL 5-10-90 DUE 11-10-90: ST-62515 CAL 5-10-90 DUE 11-10-90: L-014 DUE 11-16-90
ST-0700 CAL 12-9-89 DUE 12-9-90

Test Media: Deionized Water

Paragraph 4.2	Test Value	Requirement
Water Specific Resistance	<u>17 MΩ</u> ohms/cm	<u>500,000</u> ohms/cm min
Water pH	<u>6.5</u>	<u>5.5 to 8.0</u>
Filled per Paragraph 4.2.1	<u>YES</u>	<u>Compliance</u>

- | | Temp.
(degree F) | Weight
(lb) |
|--|---------------------|-------------------------|
| A) Empty Specimen & Fixture (W_e) | | <u>477.9</u> |
| B) Filled Specimen & Fixture, (W_f)
(No fill tube installed) | <u>72°</u> | <u>1291.8</u> |
| C) <u>TANK CAPACITY</u> = ($W_f - W_e$) X Specific Volume <u>22580</u> (cu-in) (V_c)
Specific Volume is taken from Table 1.
at measured temperature. $V_c > 22450$ cu-in <u>YES</u> complies | | |
| D) Filled Specimen and Fixture
(Fill tube installed) | | <u>1293.1</u> (W_t) |

Tested By Matt Friscia Date 5-16-90 Specimen Passed YES

Witnessed By W. P. ... Date 5-16-90 Specimen Failed NO

IAC Witnessed By ... Date 5-16-90

TRW PRESSURE SYSTEMS, INC.

TRW PSI Procedure No. 50-000374
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DATA SHEET "B"
PROOF PRESSURE & INTERNAL VOLUME
(SHEET 2 OF 2)

Customer: Hughes Aircraft Company

Date: MAY 15, 1990

Customer Part No. 4946407

TRW PSI Part No. 80350-1

Test Para. No. 4.2.2

TRW PSI Serial No. 0003

Test Equipment: E-0578 Cal 4-18-90 DUG 10-18-90: ST-0625

Cal 5-10-90 DUG 11-10-90: ST-62515 Cal 5-10-90 DUG 11-10-90: L-014 Cal 5-16-90 DUG 11-16-90
ST-0700 Cal 12-9-89 DUG 12-9-90
Filled Specimen Temperature Degree F 72°

Specimen Test Pressure	Actual Pressure Psig	Wp Weight lbs	V Delta Volume In	Total Volume Vt In	
A) 65 psig	<u>65</u>	<u>1295.8</u>	<u>91.6</u>	<u>22671.6</u>	V _{20%}
B) 130 psig	<u>130</u>	<u>1297.9</u>	<u>149.8</u>	<u>22729.8</u>	V _{40%}
C) 195 psig	<u>195</u>	<u>1300.0</u>	<u>208.1</u>	<u>22788.1</u>	V _{60%}
D) 260 psig	<u>260</u>	<u>1302.0</u>	<u>263.6</u>	<u>22843.6</u>	V _{80%}
E) 325 psig	<u>325</u>	<u>1304.1</u>	<u>321.8</u>	<u>22901.8</u>	V _{100%}
F) 65 psig (second cycle)	<u>65</u>	<u>1295.9</u>	<u>94.4</u>	<u>22674.4</u>	V _{ss}

G) $V_{ps} = .006 \times V_{100\%}$ 132.4

H) $V_{ss} - V_{20\%} < V_{ps}$ Yes complies.

Note: $V = (W_p - W_t) \times \text{Specific Volume}$. (Specific Volume is taken from Table 1. at the measured temperature.) W_t taken from Data Sheet "B", Sheet 1.

$V_t = V + V_c$, V_c taken from Data Sheet "B", Sheet 1.

Tested By Martin Frasca Date 5-15-90 Specimen Passed YES

Witnessed By Martin W. Frasca Date 5-16-90 Specimen Failed NO

HAC Witnessed By Martin E. R. Martin Date 5-16-90

Pressure Cycles

Tank is pressurized to 325, +10/-0 psid, while the pressurization rate is controlled from 0 psid to 270 psid in 5 to 7 seconds. Number of cycles is 2.

Tank is pressurized to 260, +10/-0 psid, while the pressurization rate is controlled from 0 psid to 260 psid in 5 to 7 seconds. Number of cycles is 75.

Acoustic Test

Acoustic Test Requirements

Octave (Hz)	Envelope (dB)	Tolerance (dB)	Overall (dB)	Duration (minutes)
31.5	129.1	+/- 3		
63	134.7	+/- 3		
125	140.0	+/- 3		
250	144.1	+/- 3	148.3	3
500	142.9	+/- 3	+/- 1.5	
1000	138.0	+/- 3		
2000	132.0	+/- 3		
4000	126.8	+/- 3		
8000	124.0	+/- 3		

Tank is empty and unpressurized.

Acoustic noise test is conducted for a total time of 3 minutes.

Sine Vibration (Dry)

Sinusoidal Vibration Spectrum

<u>Axis</u>	<u>Frequency</u>	<u>Acceleration</u> <u>(g - peak)</u>	<u>Displacement</u> <u>(DA)</u>
X, Y, Z	5 - 10 10 - 100	---- 2.5	13 mm (.5 in.)

The sine input shall not be notched.

Tank is unpressurized.

The sinusoidal logarithmic sweep is applied at the rate of 2 octaves/minute in each axis (X, Y, Z).


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TRW PRESSURE SYSTEMS, INC.

TRW PSI Procedure No. 50-000374
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DATA SHEET "D"
RESONANCE SEARCH & SINUSOIDAL VIBRATION


Customer: Hughes Aircraft Company Date: 5-30-90
Customer Part No. 4946407 TRW PSI Part No. 80350-1
Test Para. No. 4.5 TRW PSI Serial No. 0003
Test Equipment: SEE ATTACHED EQUIPMENT LIST

	Actual	Requirement
Specimen Test Pressure	<u>0</u> psig	<u>Empty, vented</u>
Bolt Torque	<u>90'</u> ^{FT} _{lbs} 	<u>20 Ft-lbs TANK BOLTS</u>
Visually Examine Support Structure and Specimen for Damage After Vibration	<u>✓</u>	<u>No Visual Damage</u>

Tests Complete.

Axis	Frequency From	To	D. A. Inches	G Peak	Sweep Rate Minutes/Octave	Duration Total Minutes	Date
<u>Z</u>	<u>SEE TEST SEQ # 7</u>					<u>5-30-90</u>	

Date	Time	LOG ENTRIES

Tested By George W. Casey  Date 5-30-90 Specimen Passed ✓
Witnessed By John F. Kuehler Date 5-30-90 Specimen Failed
HAC Witnessed By Sam D. Madala Date 13 Jun 90

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TRW PRESSURE SYSTEMS, INC.

TRW PSI Procedure No. 50-000374
Page 26 Revision "A"

DATA SHEET "D"
RESONANCE SEARCH & SINUSOIDAL VIBRATION

Customer: Hughes Aircraft Company Date: 5-30-90
Customer Part No. 4946407 TRW PSI Part No. 80350-1
Test Para. No. 4.5 TRW PSI Serial No. 0003
Test Equipment: SEE ATTACHED EQUIPMENT LIST

	Actual	Requirement
Specimen Test Pressure	<u>0</u> psig	<u>Empty, vented</u>
Bolt Torque	<u>90' ^{psi} / lbs</u>	<u>20 Ft-lbs TANK BOLTS</u>
Visually Examine Support Structure and Specimen for Damage After Vibration Tests Complete.	<u>✓</u>	<u>No Visual Damage</u>

Axis	Frequency From	To	D. A. Inches	G Peak	Sweep Rate Minutes/Octave	Duration Total Minutes	Date
<u>Y</u>	<u>SEE TEST SEQ # 8</u>					<u>5-30-90</u>	

Date	Time	LOG ENTRIES

Tested By George W. Payer ^{NY 15 90} Date 5-30-90 Specimen Passed ✓
Witnessed By John. Kowey Date 5-30-90 Specimen Failed
HAC Witnessed By Tom J. Heath Date 13 June 90

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TRW PRESSURE SYSTEMS, INC.

TRW PSI Procedure No. 50-000374
Page 26 Revision "A"

DATA SHEET "D"
RESONANCE SEARCH & SINUSOIDAL VIBRATION

Customer: Hughes Aircraft Company

Date: 6-1-90

Customer Part No. 4946407

TRW PSI Part No. 80350-1

Test Para. No. 4.5

TRW PSI Serial No. 0003

Test Equipment: SEE ATTACHED EQUIPMENT LIST

	Actual	Requirement
Specimen Test Pressure	<u>0</u> psig	<u>Empty, vented</u>
Bolt Torque	<u>90*</u> Ft/Lbs (FIXTURE)	<u>20 FT/LBS TANK BOLTS</u>
Visually Examine Support Structure and Specimen for Damage After Vibration Tests Complete.	<u>✓</u>	<u>No Visual Damage</u>

Axis	Frequency From	To	D. A. Inches	G Peak	Sweep Rate Minutes/Octave	Duration Total Minutes	Date
<u>X</u>	<u>SEE TEST SEQ # 14</u>						<u>6-1-90</u>

Date	Time	LOG ENTRIES

Tested By George W. Payer Date 6-1-90 Specimen Passed ✓
 Witnessed By Ty Kuan Ku Date 6-13-90 Specimen Failed No
 HAC Witnessed By Gina R. Madala Date 13 June '90

Random Vibration (Wet)

Axes	Frequency	PSID		Grms
		(G ² /Hz)	(dB/oct)	
X, Y & Z	20 - 50		+ 6	
	50 - 200	.076		
	200 - 390		- 6	6.14
	390 - 800	.020		
	800 - 2000		- 6	

NOTE: The random spectrum shall be program or active notched in narrow frequency bands at the fundamental resonance so as to achieve the response levels in each axis as follows:

X axis 10 G peak
Y axis 10 G peak
Z axis 12 G peak

The tank is filled with 1135 pounds of Freon and pressurized to 135, +5/-0 psig with nitrogen.

Vibration levels are applied for a duration of 3 minutes per axis.

Collapse Pressure

The tank is subjected to ambient external pressure (14.6 psi minimum) with simultaneous internal evacuation to $P = -4.0, +0.1/-0$ psid.

This pressure is held for 15, $+1/-0$ minutes.

Burst Pressure Test

The burst test requires pressurization up to 357 psig. The normalized burst pressure requirement is 410 psid.

The actual burst pressure was 555 psig. The normalized burst pressure was 470.9 psig.

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PRESSURE SYSTEMS, INC.

TRW PSI Procedure No. 50-000375
Page 59 Revision "B"

DATA SHEET "U"
BURST TEST

Customer: Hughes Aircraft Company Date: 10/8/90
 Customer Part No. 4946407 TRW PSI Part No. 80350-1
 Test Para. No. 4.27 TRW PSI Serial No. 0003
 Test Equipment: GAUGE: S/N 62515 #ST-0625 CALIBRATION DUE 3/5/91

Test Media: Deionized Water

	<u>Test Value</u>	<u>Requirement</u>
A) Water Specific Resistance	<u>15</u> megohms/cm	<u>0.5 Megohms/cm min.</u>
B) Water pH	<u>6.5</u>	<u>5.5 to 8.0</u>
C) Specimen (Water) Temperature	<u>72</u> deg. F	<u>N/A</u>
D) Specimen Filled	<u>YES</u>	<u>Compliance</u>
E) Pressurization Rate (less than 357 psig)	<u>60 psig / min</u> <u>sec.</u>	<u>75 psig/min.</u>
F) Pressurization Rate (above 357 psig)	<u>60 psig / min</u> <u>sec.</u>	<u>1000 psig/min.</u>
G) Rupture Pressure	<u>555</u> psig	<u>Record</u>
H) Burst Pressure	<u> </u> psig	<u>(normalized)</u> <u>410 psid minimum</u>

Tested By: [Signature] Date 10/8/90 Specimen Passed YES
 Witnessed By: N/A (0/8) Date 10/8/90 Specimen Failed NO
 HAC Witnessed By: [Signature] (SCG VCERT 402) Date 10-8-90

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